

Analysis of Structural Dynamics Turbo Generator Load on Foundation Structure for Estimation Stresses during Vertical Excavation Using Finite Element Method

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Abstract

The load over the turbo generator should be visualized .Based on the finite element value and experimental value of soil parameter as such major principal stresses and minor principal stresses, normal stresses, shear stresses, bearing pressure were analyzed for the analysis of foundation structure .The analysis of factor bending moment, intensity of pressure, maximum pressure , main area of steel ,distribution of steel were carried out for a comparative statement of the two dimensional and three dimensional approach .Three dimensional approach became realistic and exact approach over two dimensional approach .

1. Introduction to Structural Dynamics Turbo Generator Foundation Load

Statics load ,Beam weight $Q = 9L = 1.2 \times 7.9 \times 24 = 225.72 \text{ KN}$

Machine weight $p = 700 \text{ KN}$,Load due to longitudinal girder $N1 = 1.2 \times 1.0 \times 65 \times 24 = 18.72 \text{ kN}$

$1/3^{\text{rd}}$ self weight of the column , $N2 = .8 \times 24 \times 9.3/3 = 59.52 \text{ Kn}$

$H1 = 12.0 - 1.2 - 1.5 = 9.3 \text{ m}$, $N = N1 + N2 = 78.24 \text{ KN}$, $l = 1/12 \times 1 \times (1.2)^3$, Effective span and height of the frame $Lo = 3.4 \text{ m}$

$Z = .6$, $h0 = 9.9 \text{ m}$, $Z = .6 \text{ m}$, $X = .5 \text{ m}$,

$h0/10 = 9.9/3.4 = 2.917$, $x/10 = .5/3.4 = .14$, $\alpha = .01$, Effective span , $l = 10 - 2 \times \alpha = 3.4 - 2 \times .5 \times .01 = 3.39 \text{ m}$

Effective height $h = h0 - z \alpha = 9.9 - .6 \times .01 = 9.798 \text{ m}$, $K = (Ib/Ic)h/l = 5.293$

Deflection due to concentrated load $P = (Pl / 06Ef Ib) * (2k+1)/k+2$
 $= (700 * 3.39 / 96 * 3 * 10^7 * .144) * (2 * 5.293 + 1) / (2 * 5.293 + 2) = 6.063 \text{ E-5 m}$

The deflection due to udl at beam center

$$=(Ql^3/384EfIb)x(5k+2)/(k+2)=$$

$$225.72*3.39/384*3*10e7*.144(5*5.293+2)/(5.2932+2) =4,2910e-6 m$$

Deflection at the center of the beam due to shearing forces

$$= 3/5(L/Ef Ab(P+Q/2) =3/5X(3.39/3X10E+7X1.2)X(700+225.72/2) =39.17X10E-7m$$

The axial compression of the column

$$= (h/Ef Ac)*(N+ (P+Q)/2) =(9.8/3*10e7*1.0)*(43.04+(700+225.72)/2) =61.29*10e-7m$$

Total deflection at center of the beam = $\Delta 1+\Delta 2+\Delta 3+\Delta 4$

$$=147.05*10e-6 m$$

Vertical natural frequencies of the frame A1 is (fn) $A1= 30/(\Delta)^{1/2}$

Frame A2 ,Beam cross section area ,Ab = 1.4mx1.2m =1.687m² ,Column cross section area =1.0*1.4 =1.4 m²,Ib= 1/12)*1.4*1.2³ =0.2016 m⁴,Ic=1/12*1.4*1.0³= .116 m⁴ ,K= 2.322,l=3.2m,h=4.3 m,static load ,P= 900 KN ,Q= 91= 96.7 KN ,N=(.85+.65) * 1.2 *24 + 1/3(1.4*24*3.8) = 85.76

$$Kn,\Delta 1=(900*3.2^3/96*3*10e+7*.2016)(2*2.322+1)/4.322 =66.33*10e-6$$

$$m,\Delta 2=(96.76*3.2^3/3.84*3*10e7*.2016)(5*2.322+2)/2.322+2 =4.3*10e-6 m ,\Delta 3 =$$

$$3/5(3.2/310e7*1.68(900 + 96.76/2))=36.12*10e-6,\Delta 4=$$

$$4.3/3*10e7*1.68(85.76+(900+96.76)/2), Total deflection at center of the beam =$$

$$\Delta 1+\Delta 2+\Delta 3+\Delta 4 =166.55*10e-6 m,(fn)A2= 30/\Delta^{1/2} = 2324.6 rpm,Frame A3,Ab= 1.44$$

$$m^2,Ac = 1.2 m^2,Ib= 1/12(1.2*1^3) = .1 m^4,K= 2.322,Loads P= 600KN,Q=$$

$$ql=1.44*2.4*24=82.94 Kn,N=.85*1.2*1*24+1/3*1.2*24*3.8 =60.96 KN,\Delta 1=$$

$$(600*3.2^3/96*3*10e7*.1728)(2*2.322+1)/4.322 =51.59*10e-6m,\Delta 2=$$

$$82.94*3.2^3/384*3*10e7*1.44(5*2.322+2/2+2.322) =28.5*10e-6m,\Delta 3=$$

$$3/5*3.2/3*10e7*1.44(600+82.94/2) =28.5*10e-6$$

$$m,\Delta 4=4.3/3*10e7*1.2(60.96+682.94/2) =48.06*10e-6 m,\Delta =132.44*10e-6 m ,(fn)An=$$

$30/(\Delta)^{1/2} =2606.82 rpm$,Fnh=horizontal natural frequency,Khi = lateral stiffness of cross frame I of in ton /meter,Wi =total weight on the frame I including machine weight ,weight transevers beam and weight transmitted by the longitudinal girder.

$IG = \sum w_i * G_i^2 = WA * GA^2 + WB * GB^2$,XGi = Distance of weight Wi from the vertical axis through the center of gravity ,IH = $\sum k_{hi} * H_i^2 = KhA * HA^2 + KhB * HB^2 + KhC * HC^2 + KhD * HD^2$

XHi= distances of cross frame I from the center of rigidity H , $\alpha 0=1/2(e^2 \sum k_{hi}/IG + \sum k_{hi}/\sum W_i + I H/IG)$.

Horizontal frequencies, $W_i = P_i + +QI + 2 N_i$,WA1 = 700 +69.12+ 2* 43.04 = 855.20 KN

WA2= 900+ 96.76+2*85.76=1168.28 kN,WA3= 600+ 82.94+ 2*60.96=804.86 KN

$XG = (1168.28*2.5+804.86*5.5)/ 2828.34$,XG =2.6m,K= 2.322m ,XG = Disatance of resultant from the axis of the frame ,(Δh)A1 = $h^3/12Ef Ic(3k+2)/6k+1 + 6h/5EfAc(1+Ach*18k^2/ (Abl(6k+1)^2) + h^3/Ef Ac l^2* (18k^2/ (Abl(6k+1)^2))$, $(4.3^3/(12*3*10e7*0.0833)*(3*2.322+2)/(6*2.322+1)^2)+6*4.3/5*3*10e7$

Horizontal Amplitude :Higher horizontal frequency , $R = \sum (R_b + R_c) : \sum C_n = 0.2$
 $\sum I (R_b + R_c) (f_n/f_m)^2 = 2 * 255 * (776.6/3000)^2 = 3.42 \text{ KN}$, Lateral deflection = $\sum Ch$
 $/ \sum Kh = 3.42/1878800 = 1.82 * E^{-6} \text{ m}$, Amplitude (a_n) = $14.28 * E^{-6} \leq 0.04 \text{ mm}$

Dynamic Forces, Vertical dynamic forces for Frame A1 with corrected frequencies f n

$f_n' = f_n * 1.2$, $(f_n') \Delta I = 2968.6 \leq 3000$, Dynamic Forces are taken as under, $F_b = 16 R_b$
 $(f_n'/f_n)^2$

Frame -A1, $(F_b) A_1 = 16 * 35 * (2968.6/3000)^2 = 548.3 \text{ KN}$: Frame -A2, $(F_b) A_2 = 16 * 70 * (1.2 * 2324.6/3000)^2 = 968.3 \text{ KN}$: Frame A3, $f_n' = 1.2 * 2606.82 = 3128.1$, $(f_n) A_3 = 3128.1 / 3000$

$(F_b) A_3 = 2 F_{max} / [\{ 1 - (f_m/f_n) + (f_n/f_n')^2 (\Delta/I)^2 \}]^{1/2} = 2 * 50 / [\{ 1 - (3000/3128.1) + (3000/3128.1)^2 (.4/I)^2 \}]^{1/2} = 684.4 \text{ KN}$, **Horizontal Dyanamic forces** , $F_n = 16 \sum (R_b + R_c) (f_n h'/f_m)^2 = 16 * 255 * (1.2 * 7877.6/3000)^2 = 394.7 \text{ KN}$, $(F_n)_i = F_h * K_{hi} / \sum K_{hi}$, $(F_n) A_{1i} = 394.7 * 510980 / 1878800_i$, $(F_n) A_2 = 394.7 * 510980 / 1878800_i$, $(F_n) A_3 = 132.5 \text{ Kn}$

Design of Cross Frame ,Frame A1 ,Moment Due to concentrated load P= 700 KN, $K = I_b/I_c = 2.322$, $M_A = M_D = Pl/8(k+2) = 700 * 3.2/8 * 4.972 = 64.78 \text{ KNm}$, $M_B = M_C = -2M_A = -129.56 \text{ KNm}$

Maximum Moment at beam center , $M_p = P'/4 + M_B = 430.4 \text{ KN-M}$: Shear forces , $Q_B = Q_C = 250 \text{ KN}$, **Moment due to Uniformly distributed load** , $Q = q * L = 69.12 \text{ KN}$, $M_A = M_D = ql^2/12(k+2) = 69.12 * 3.2/1284.322 = 4.26 \text{ KN m}$: $M_B = M_C = -2M_A = -8.53 \text{ KN m}$, $M_P = ql/8 + M_B = 19.12 \text{ KN m}$, $Q_C = Q_B = 34.56 \text{ KN}$, **Moment due to vertical Dynamic** , $v = 548.3 \text{ KN}$, $M_A = M_D = F/50.7 \text{ KNm}$, $M_B = M_C = -2M_A = F/101.4 \text{ KNm}$, $M_p = (548 * 3.2/8 - 101.4) = 117.92 \text{ KN m}$, $Q_B = Q_C = F/274.1 \text{ KN}$,

Moment and shear due to Horizontal Dynamic force for (Fn) A1 = 107.3 KN , $M_A = \frac{1}{2} * F_n * (3k+1)/(6k+1) = F(1/2) * 107.3 * 4.3 * [3 * (2.322+1)/6 * (2.322+1)] = F/123.1 \text{ KN m}$: $M_D = -M_A = F/123.1 \text{ KN m}$, $M_B = F_n * h/2 * 3k/(6k+1) = 107.6 \text{ KN m}$, : Shear Force $Q_B = -2M_B/l = F/67.2 \text{ KN}$, $Q_C = -Q_B = F/67.2 \text{ KN}$, **Moment and shear due to earth quake forces** , Assuming that the site is located in zone iv and seismic coefficient = 0.65 , Increase it by 50 % (as per IS Code 1893 -1970 , The coefficient become 0.075 , Lateral earth quake force = $0.075 * 2828.34 = 212.1 \text{ KN}$

$M_A = -M_D = F/243.3 \text{ KN-m}$, $M_B = -M_C = F/212.6 \text{ KNm}$, $Q_B = F/132.8 \text{ KN}$: Temperature Effect , $M_B = M_C = -3Eflb \sum T/h(K+2)$, $T = 20^\circ$ (assumed design value for the temperature) , \sum = thermal expansion coefficient = $11 * E^{-6}/^\circ \text{ C} = -3 * 3 * E^{-7} * .144 * 11 * e^{-6} * 20/4.3 * 4.322 = 153.41 \text{ KN-m}$, $M_A = M_D = -M_B(k+1)/k = 152.41(3.322/2.322) = 219.4 \text{ KN-m}$ Shrinkage Load due to shrinkage equivalent to -10° c .

2. Conclusion for Bending Moment chart and worst possible combination

Moment due to	MA KN-m	MB KN-m	Mp KN-m	Remarks
P(Vertical load)	64.78	-129.56	430.04	
Q,UDL	4.26	-8.53	19.12	
V(Vertical Dynamic load)	± 5±0.7	± 101.4	± 117.92	
H(Horizontal Dynamic load)	± 123.1	± 107.6	0	
Earth quake	± 243.3	± 212.6		
Differential temperature	219.4	-163.4	-0153.4	
Shrinkage	-109.7	76.7	76.7	
Worts possible combination				
Maximum	654.8	-611.7	643.7	
Minimum	407.06	258.8	177.8	

3. Result of Amplitude and Frequencies

Amplitude –Vertical in m	10.79*E-6	13.16*E-6	12.67*E-6	
Frequencies in rpm	2473	2324	2606	

Worst Possible combination of Bending Moment

End of Portal – Fixed	Bending Moment at A and D	Bending Moment B and C	Bending Moment At P
Maximum Bending Moment	+475 KN-m	- 492.256 KN-m	692.1`62 KN-m
Minimum Bending Moment	- 88.588 KN-m	+ 474.62 KN-m	-121.731 KN-m

References

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- [2] Dr Hassanein F. Hassan,Saif M. Jawed , "Finite element analysis of earth retaining structure problems in non linear partially saturated soil, "in the year of 2012.

